

PLAN

MATERIAL STUDIES FOR PULSED HIGH-INTENSITY PROTON BEAM TARGETS

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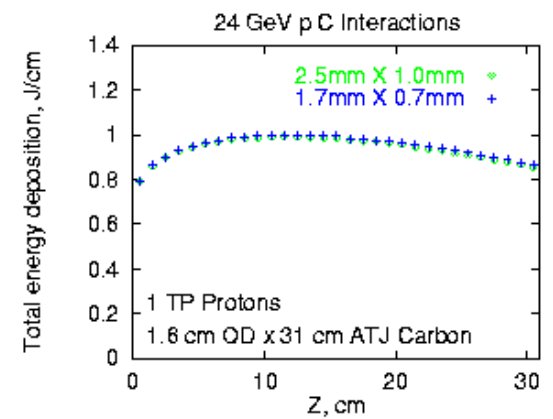
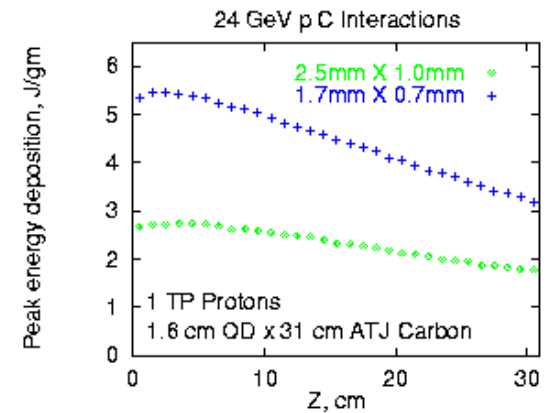
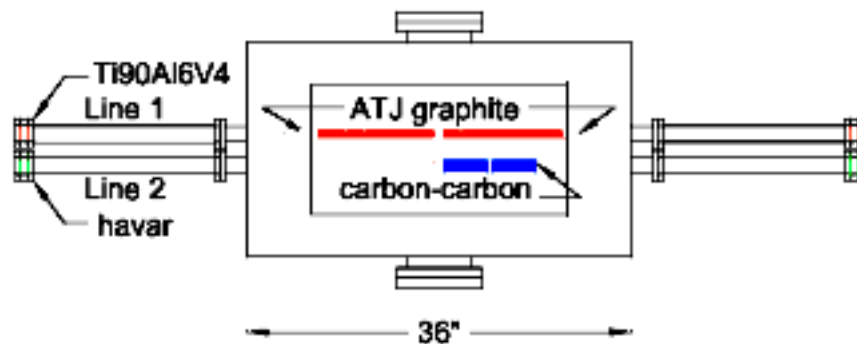
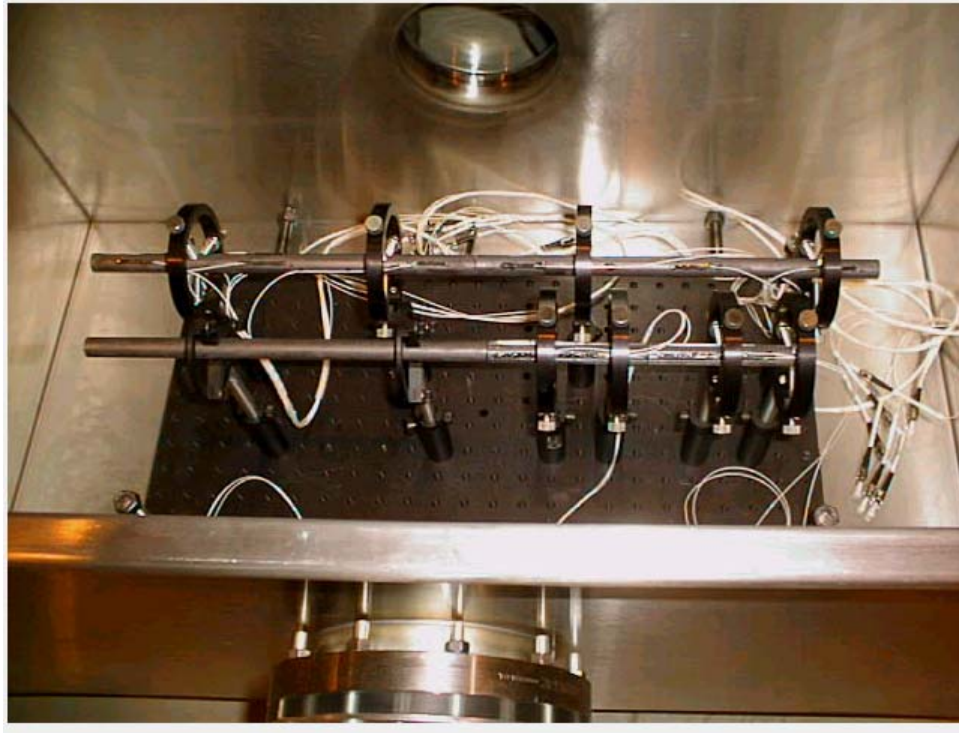
CHALLENGES FOR THE INTEGRATED TARGET SYSTEMS

AS WE GET TO 1+ MW SYSTEM

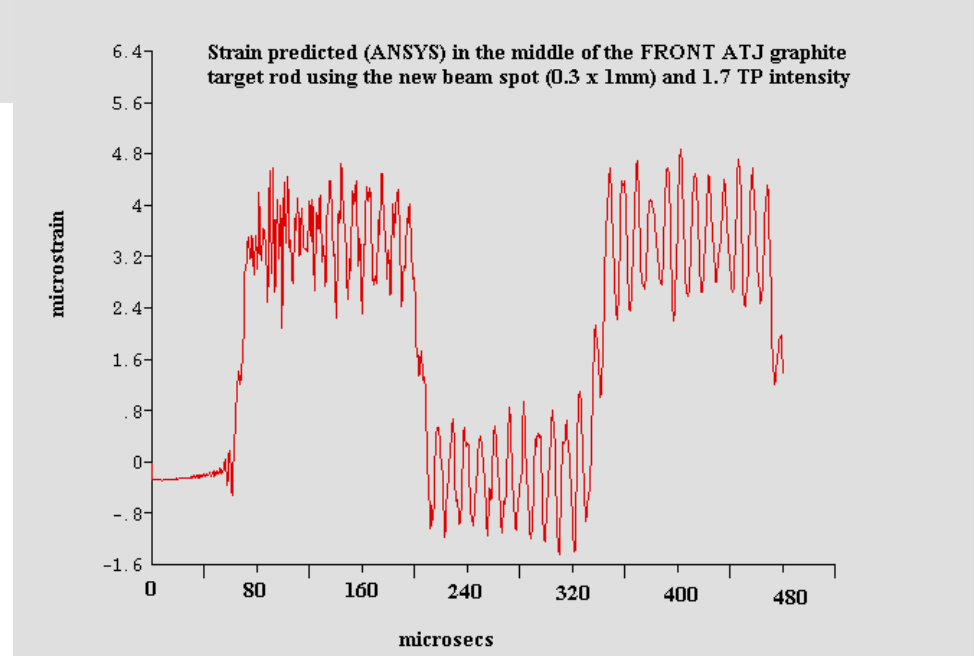
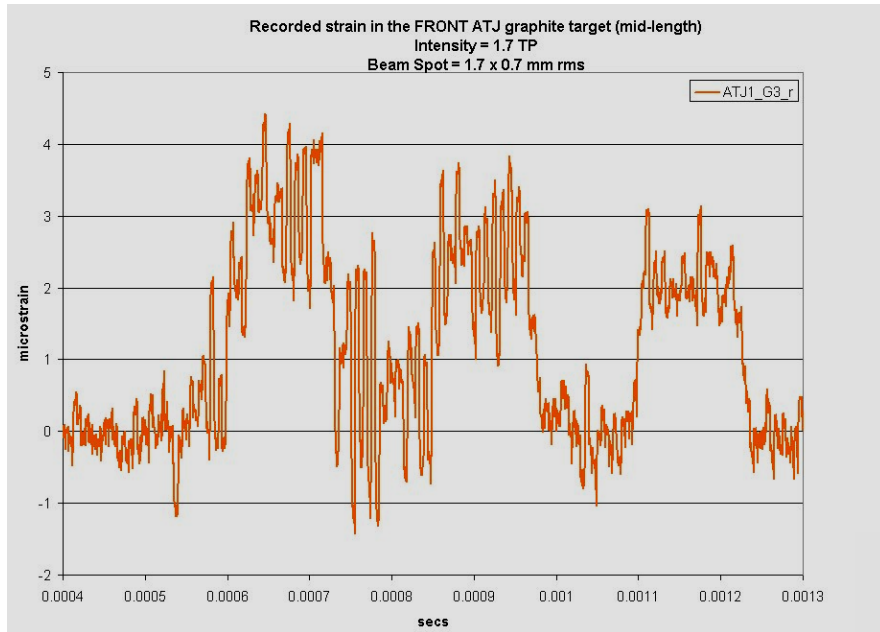
- Heat generation and removal from the target system
- Target thermo-mechanical response from energetic, high intensity protons
- Irradiation and corrosion effects on materials
- For Neutrino superbeam initiative, horn mechanical response and long term integrity (irradiation, corrosion and thermal fatigue)
- Beam windows

E951 Target Station Set-Up

Graphite & Carbon-Carbon Targets

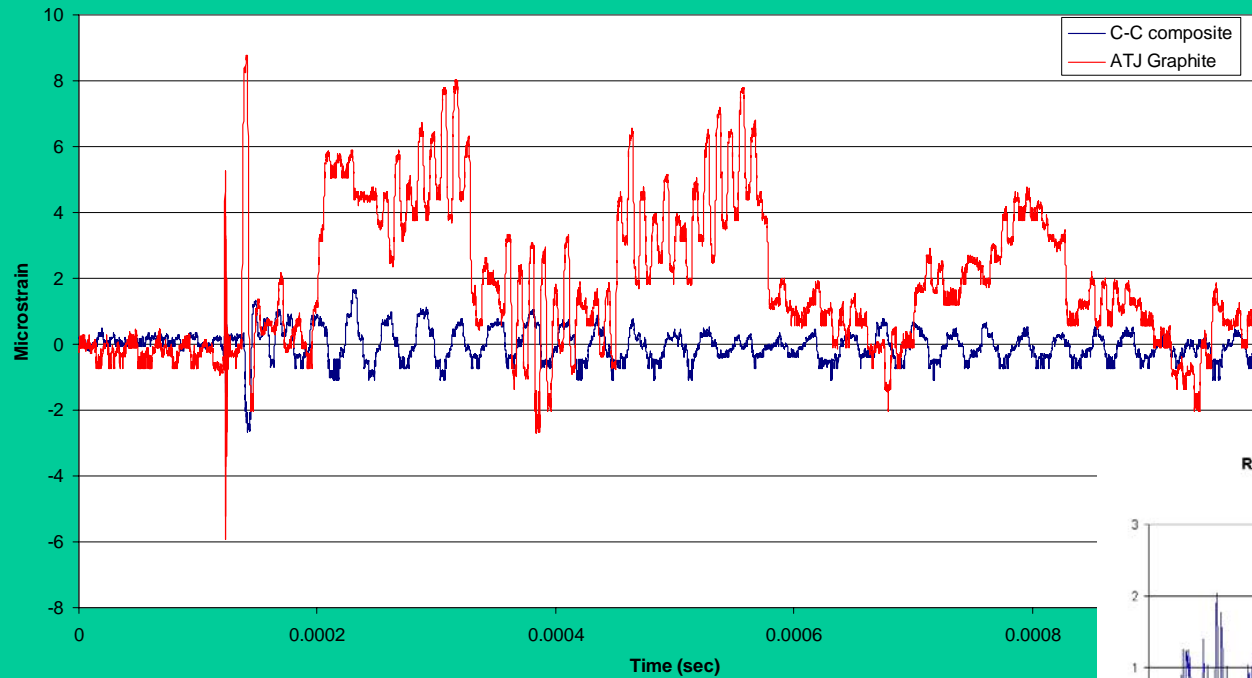


ATJ Graphite Strain Comparison

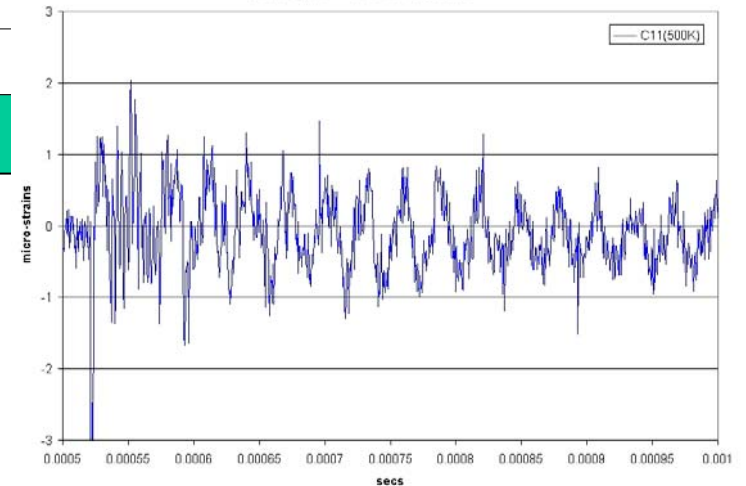


ATJ Graphite vs. Carbon-Carbon Composite

BNL E951 Target Experiment
24 GeV 3.0 e12 proton pulse on Carbon-Carbon and ATJ graphite targets
Recorded strain induced by proton pulse

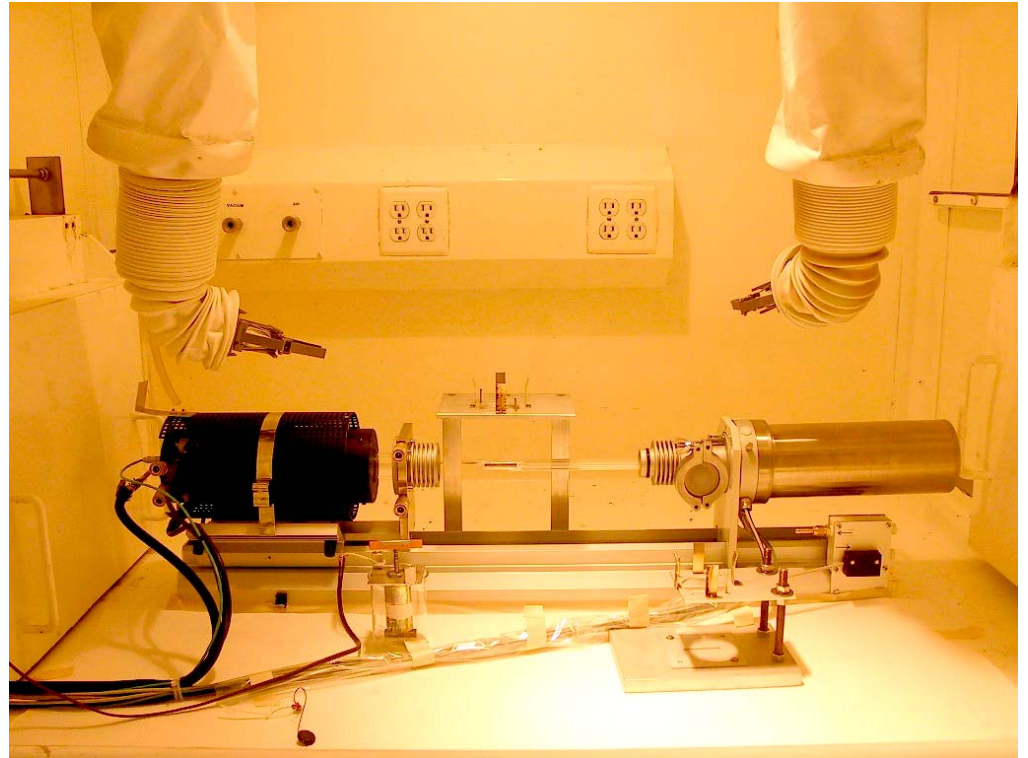
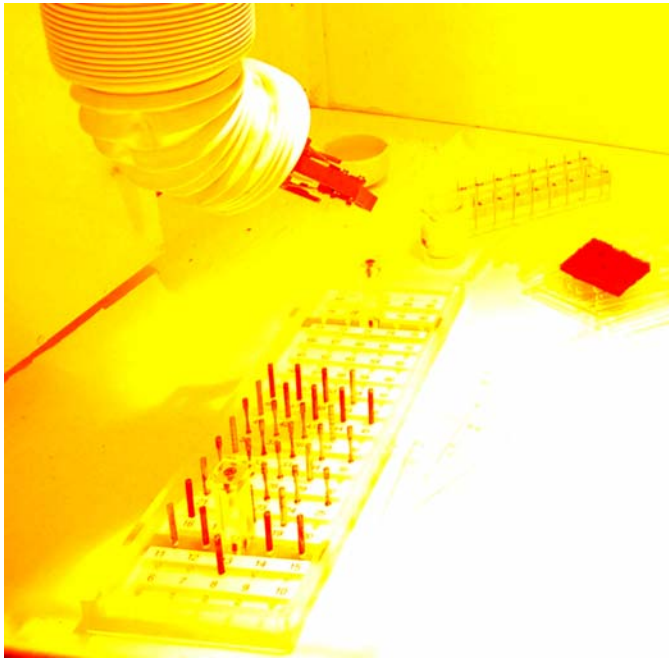


Recorded strain (500 KHz) in the FRONT C-C Target Rod
Intensity = 1.6 TP
Beam Spot = 1.7 x 0.7 mm rms



PHASE-I of BNL Irradiation Studies

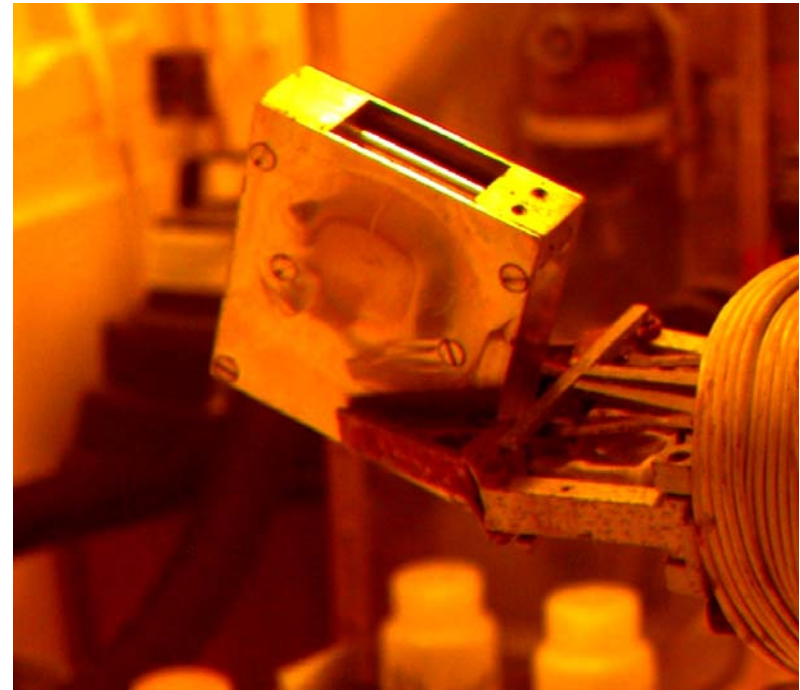
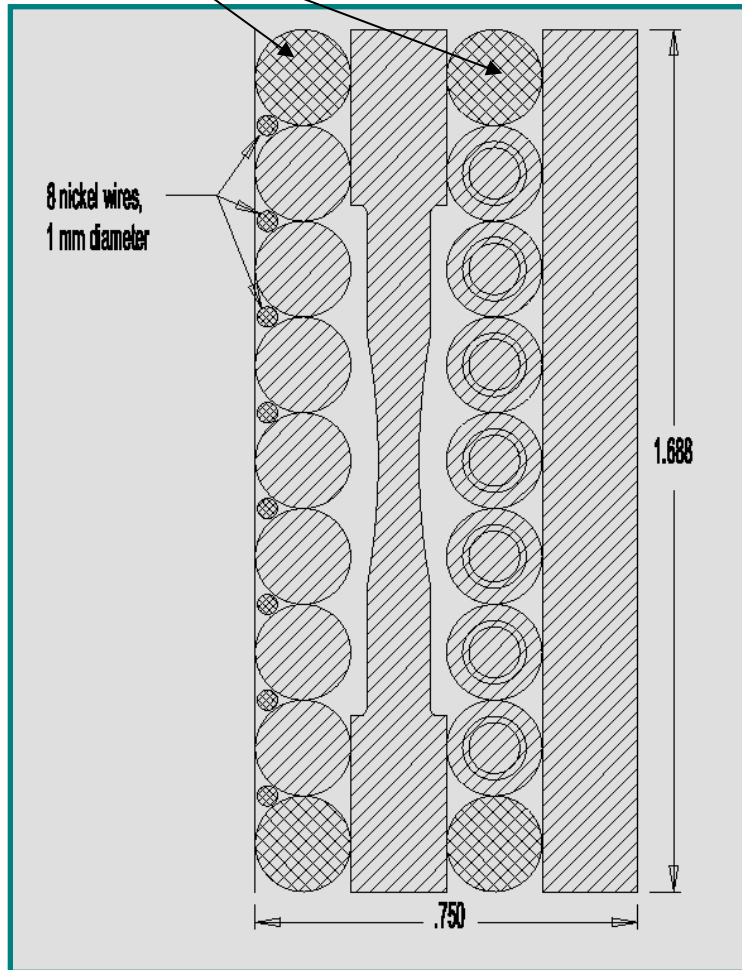
Dilatometer

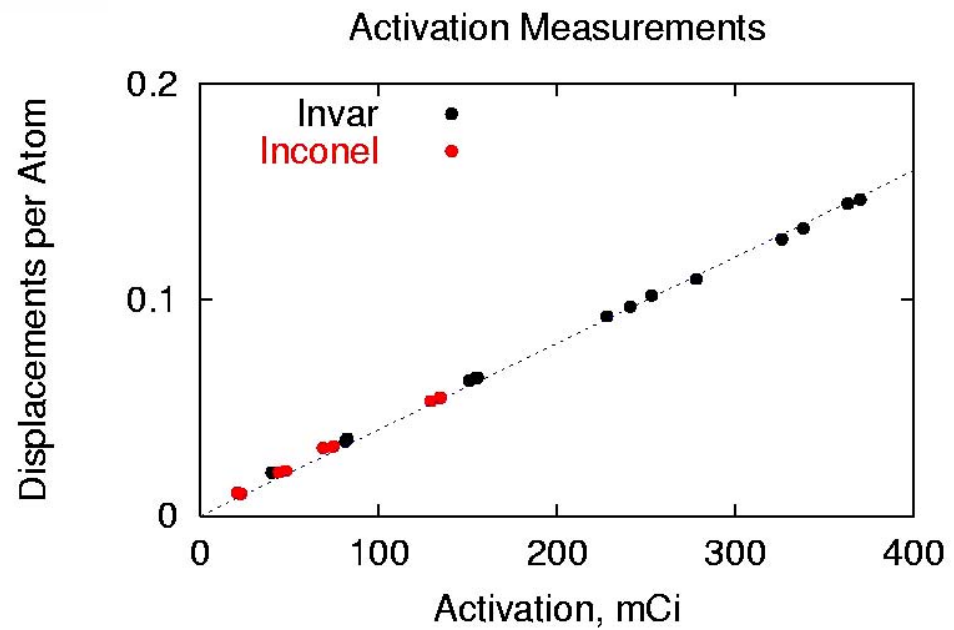
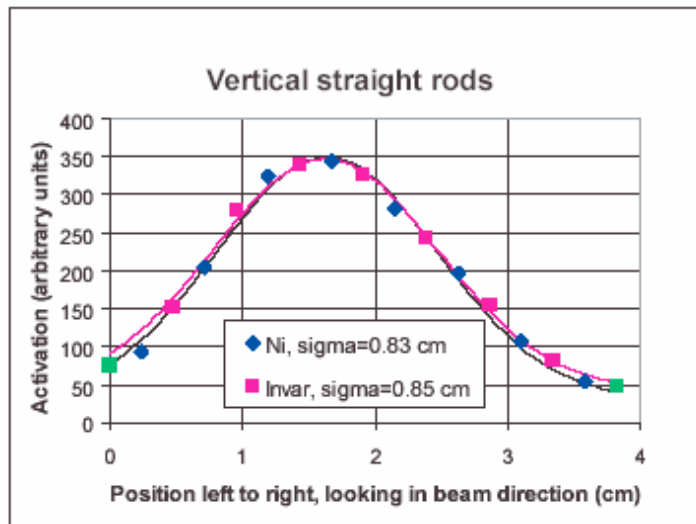


PHASE-I of BNL Irradiation Studies

Super Invar & Inconel-718

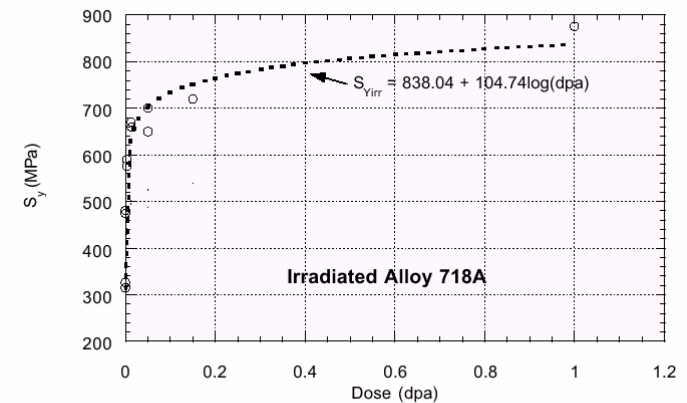
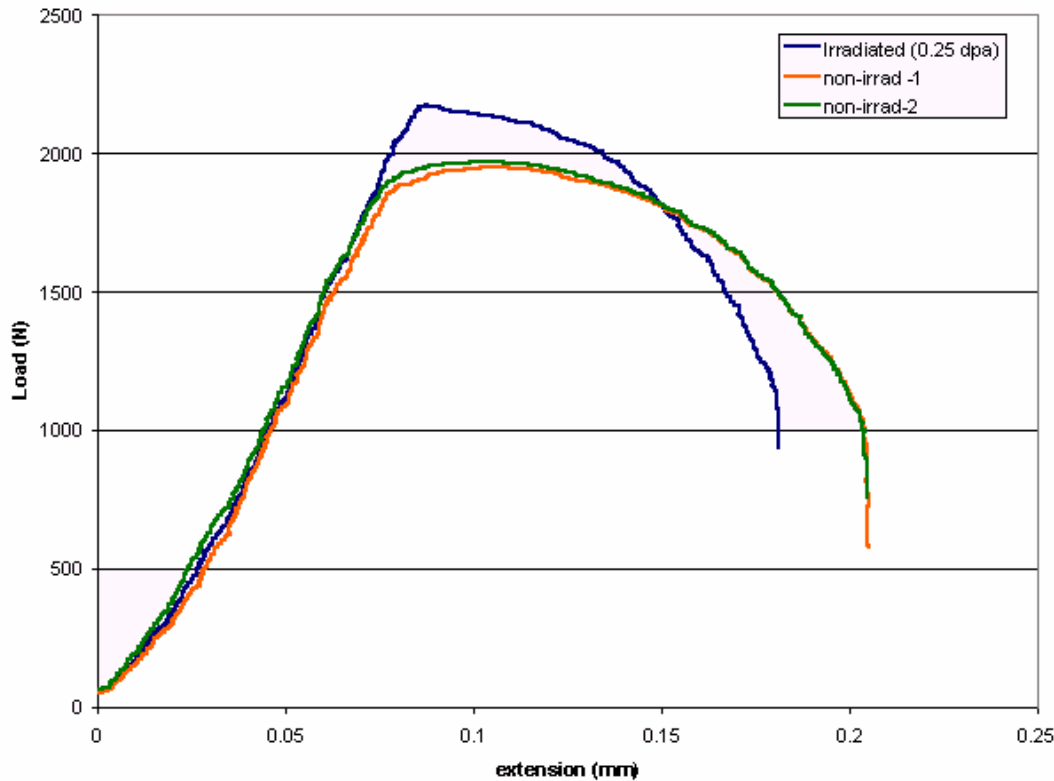
Inconel-718 rods





Super-Invar Irradiation Study

Effects of Irradiation as Observed in the Load-Displacement Curve

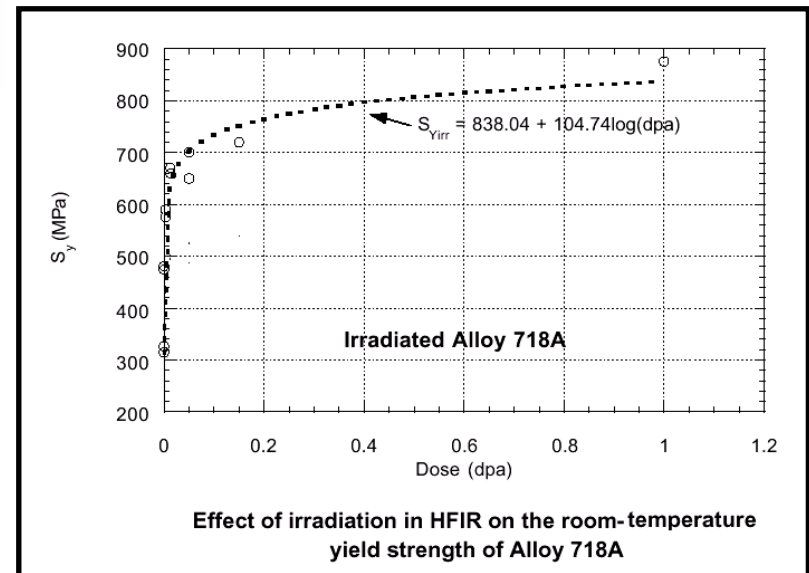
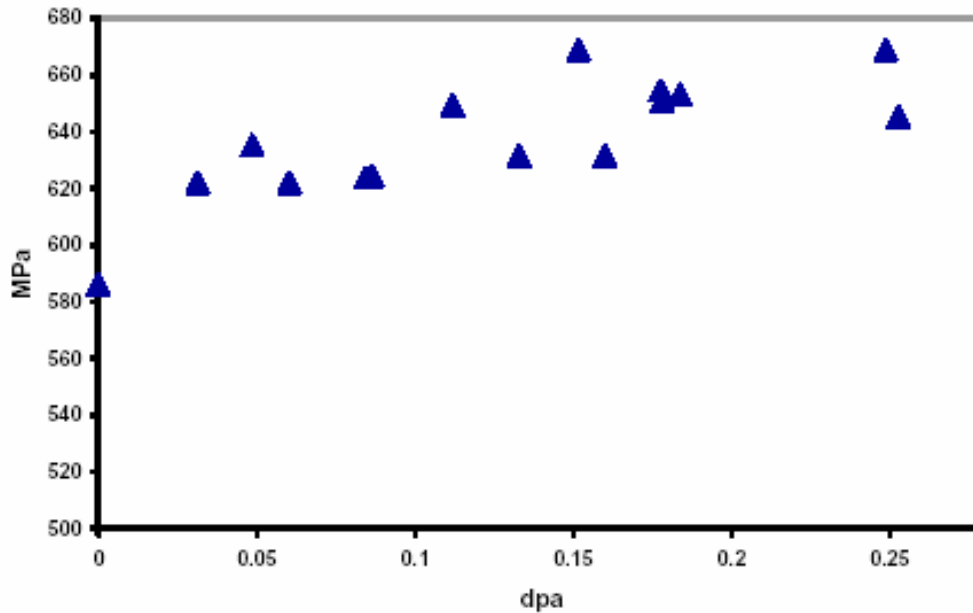


Effect of irradiation in HFIR on the room-temperature yield strength of Alloy 718A

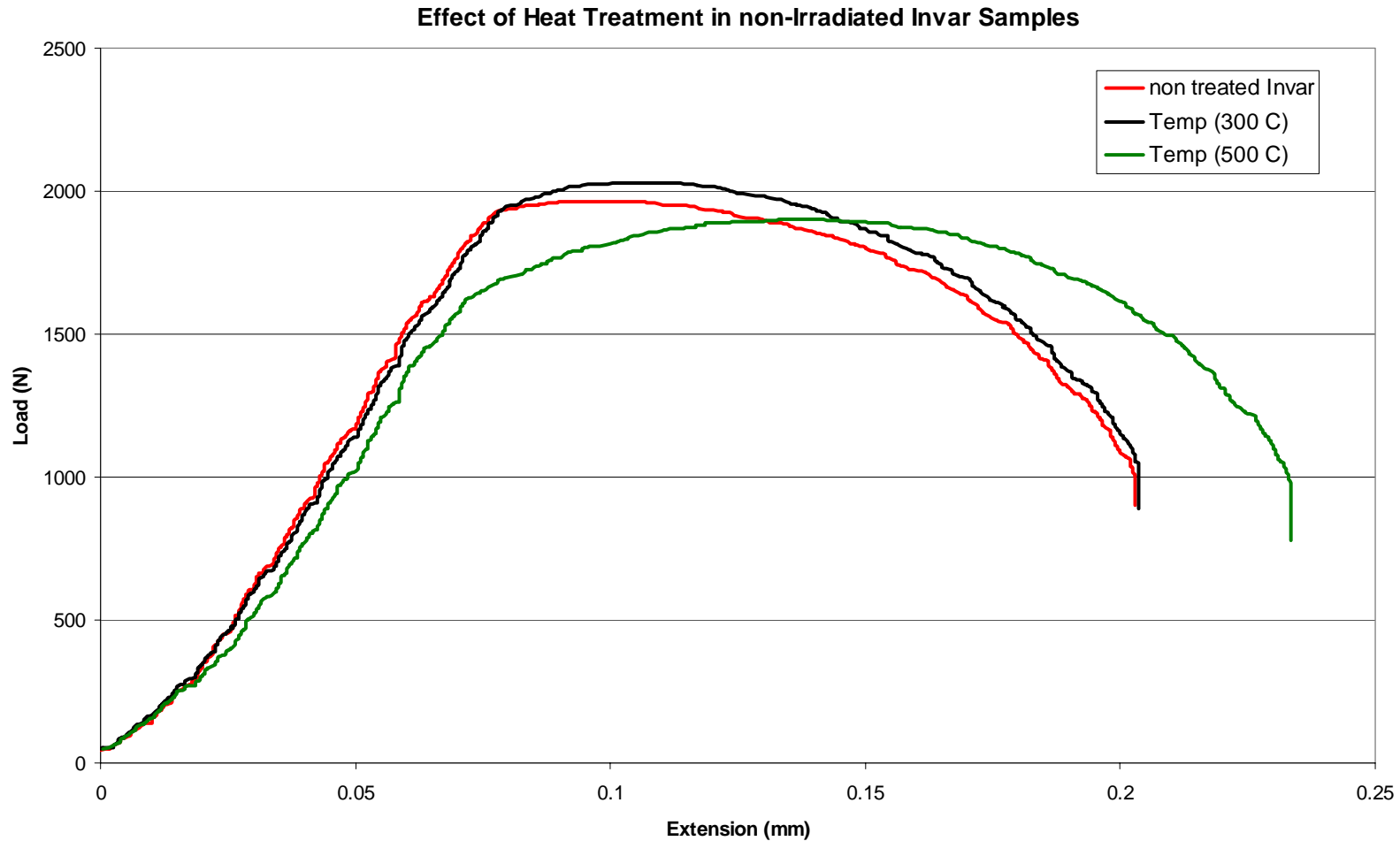
WHY STUDY super Invar ?

- High-Z with low CTE (0-150 °C)
- How is CTE affected by radiation?
- What happens to other important properties?

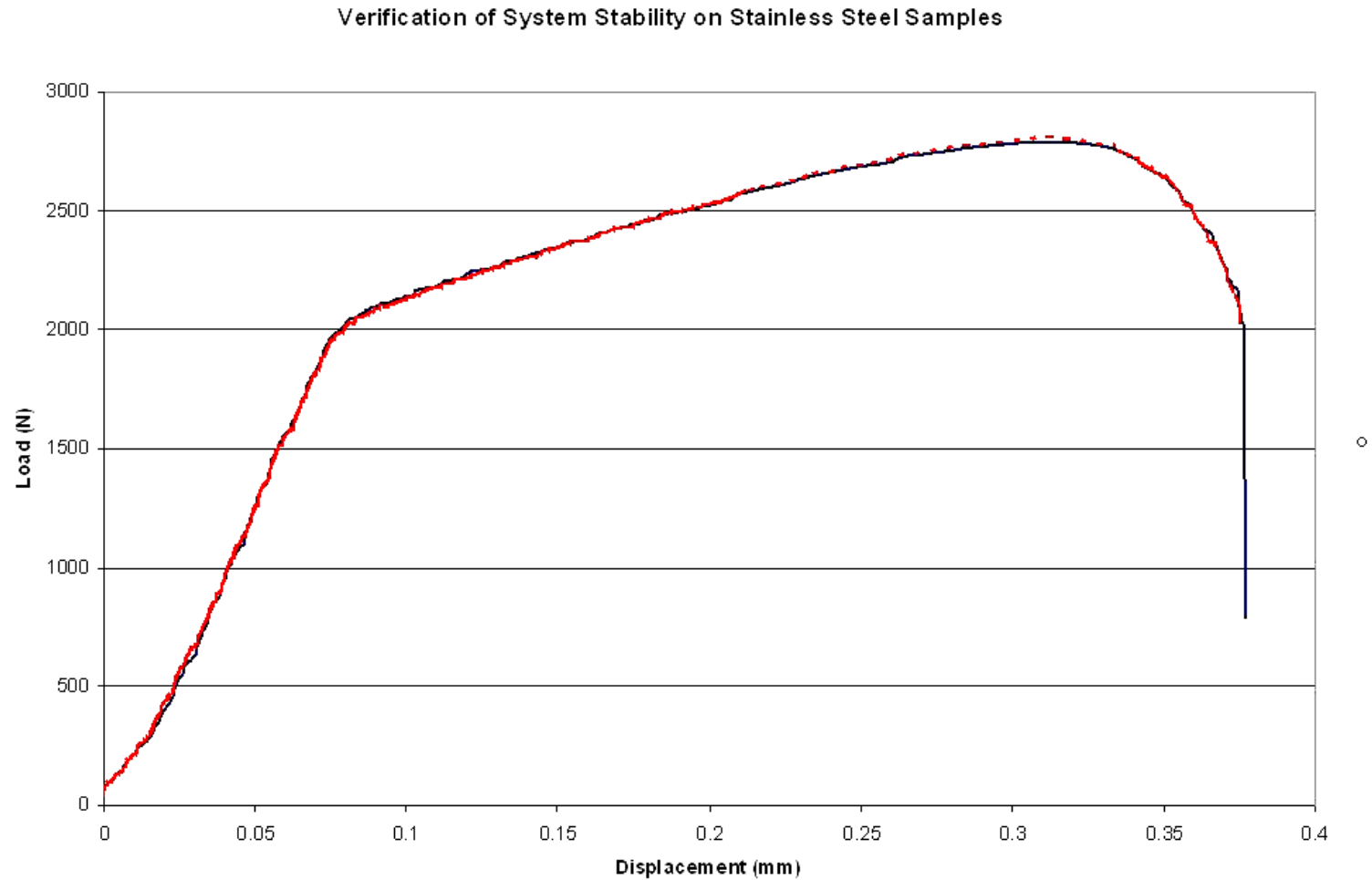
Solid Target Option: Super-Invar Irradiation Study



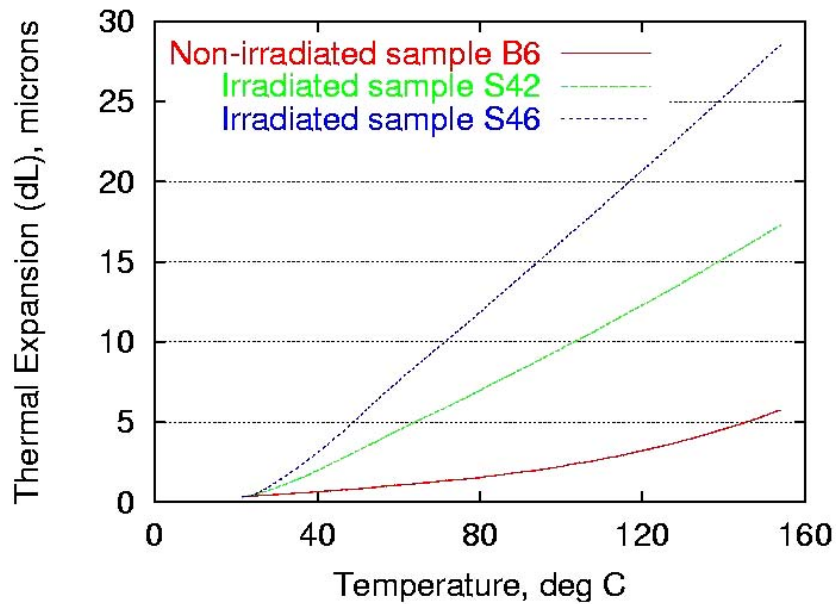
Super-Invar Irradiation Study – Temperature Effects



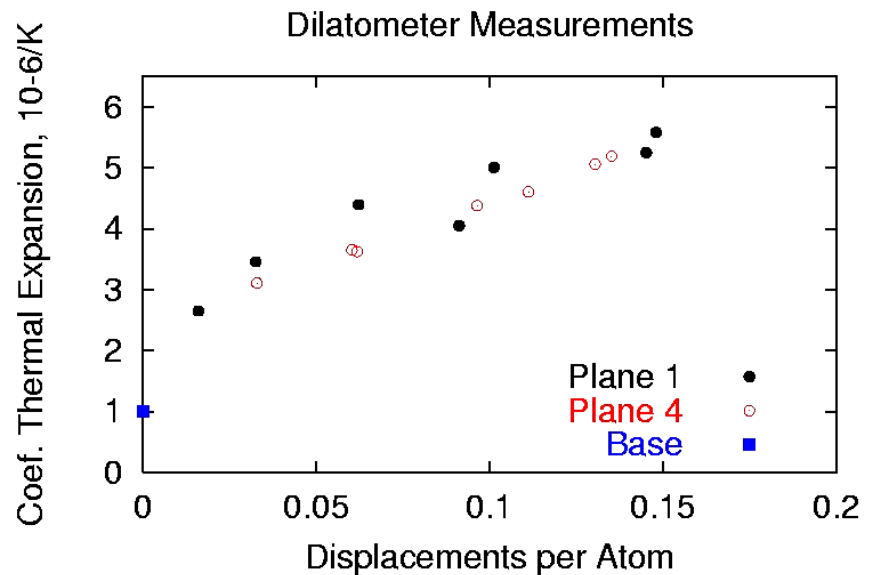
Super-Invar Irradiation Study – Temperature Effects



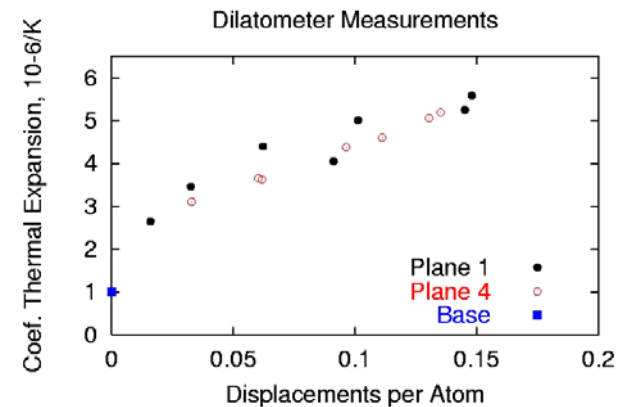
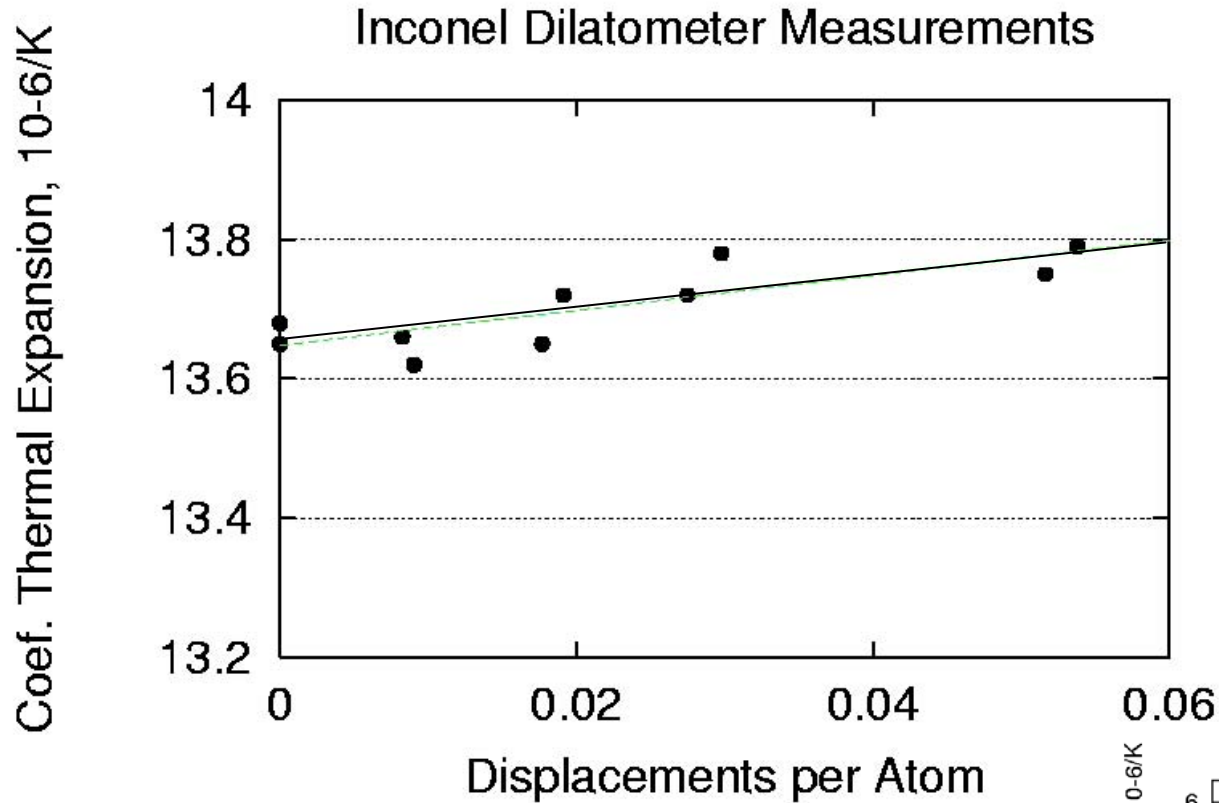
Super-Invar Irradiation Study – CTE assessment



Super-Invar



Inconel-718 CTE assessment



PHASE-II TARGET MATERIAL STUDY

WHAT'S NEXT ? Repeat irradiation/mechanical property changes experiment for baseline materials

Carbon-Carbon composite

This low-Z composite gives the indication that it can minimize the thermal shock and survive high intensity pulses. Because of its premise it is the baseline target material for the BNL neutrino superbeam initiative. The way its key properties (such as CTE or strength) degrade with radiation is unknown.

Titanium Ti-6Al-4V alloy

The evaluation of the fracture toughness changes due to irradiation is of interest regarding this alloy that combines good tensile strength and relatively low CTE

Toyota “Gum Metal”

This alloy with the ultra-low elastic modulus, high strength, super-elastic like nature and near-zero linear expansion coefficient for the temperature range -200 °C to +250 °C to be assessed for irradiation effects on these properties.

VASCOMAX

This very high strength alloy that can serve as high-Z target to be evaluated for effects of irradiation on CTE, fracture toughness and ductility loss

AlBeMet

A low-Z composite that combines good properties of Be and Al. Effects of irradiation on CTE and mechanical properties need to be assessed

PHASE-II TARGET MATERIAL STUDY

WHAT'S DIFFERENT FROM PHASE-I?

~ 100 MeV of Proton Beam (200 to 100 MeV)

Challenge of inducing UNIFORM Beam degradation

MORE Material to go in (optimization of dE/dx for range 200 MeV-100 MeV)

OPEN Issue: Study of Fracture Toughness for some of materials

Carbon-Carbon Composite Target

Temp.	% elongation
23 ° C	0%
200 ° C	-0.023%
400° C	-0.028%
600° C	-0.020%
800° C	0%
1000° C	0.040%
1200° C	0.084%
1600° C	0.190%
2000° C	0.310%
2300° C	0.405%

Gum Metal (Toyota Ti alloy)

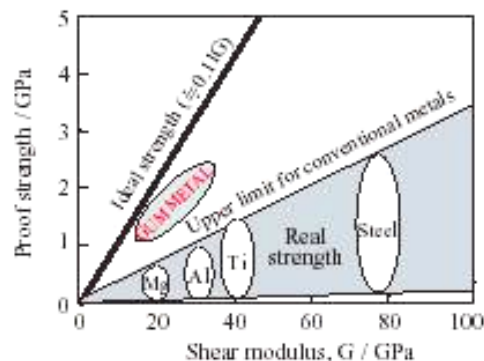


Fig. 1 Position of elastic modulus and strength of GUM METAL.

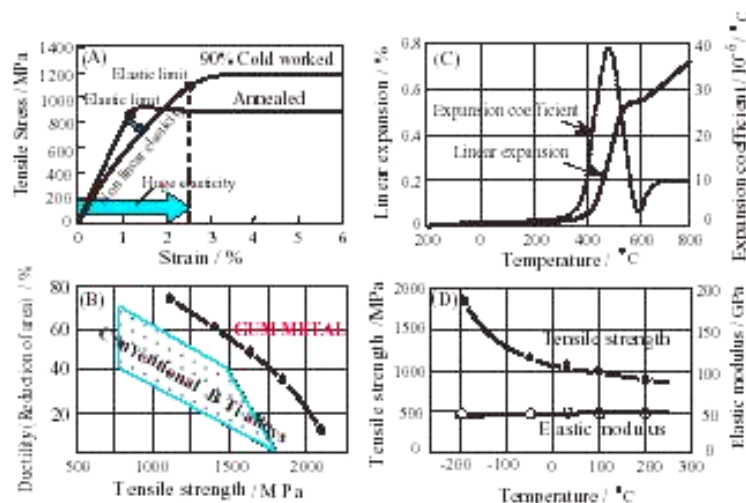


Fig. 2 Typical properties of GUM METAL.
(A) Stress-strain curve, (B) Relation between strength and ductility, (C) Invar and (D) Elinvar property.

GUM METAL named after its various unique characteristics unlike other metals is a new material having infinite possibilities in practical applications. In addition to its already commercialized applications including spectacle frames and precision screws, it is considered to be applicable in a wide range for automotive parts, medical equipments, sporting goods, decorative materials and in the aerospace industry.

References

- 1) Saito, T., et al. : Science, **300**(2003), 464
(Report received on Aug. 18, 2003)

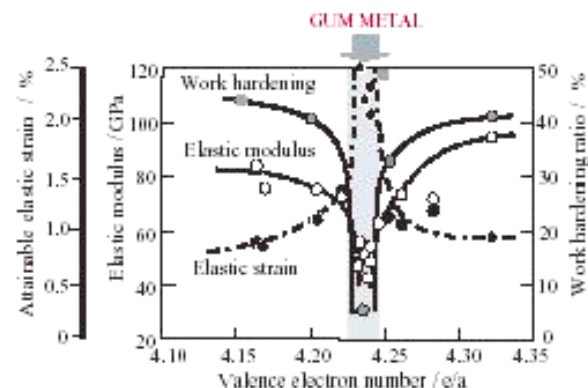
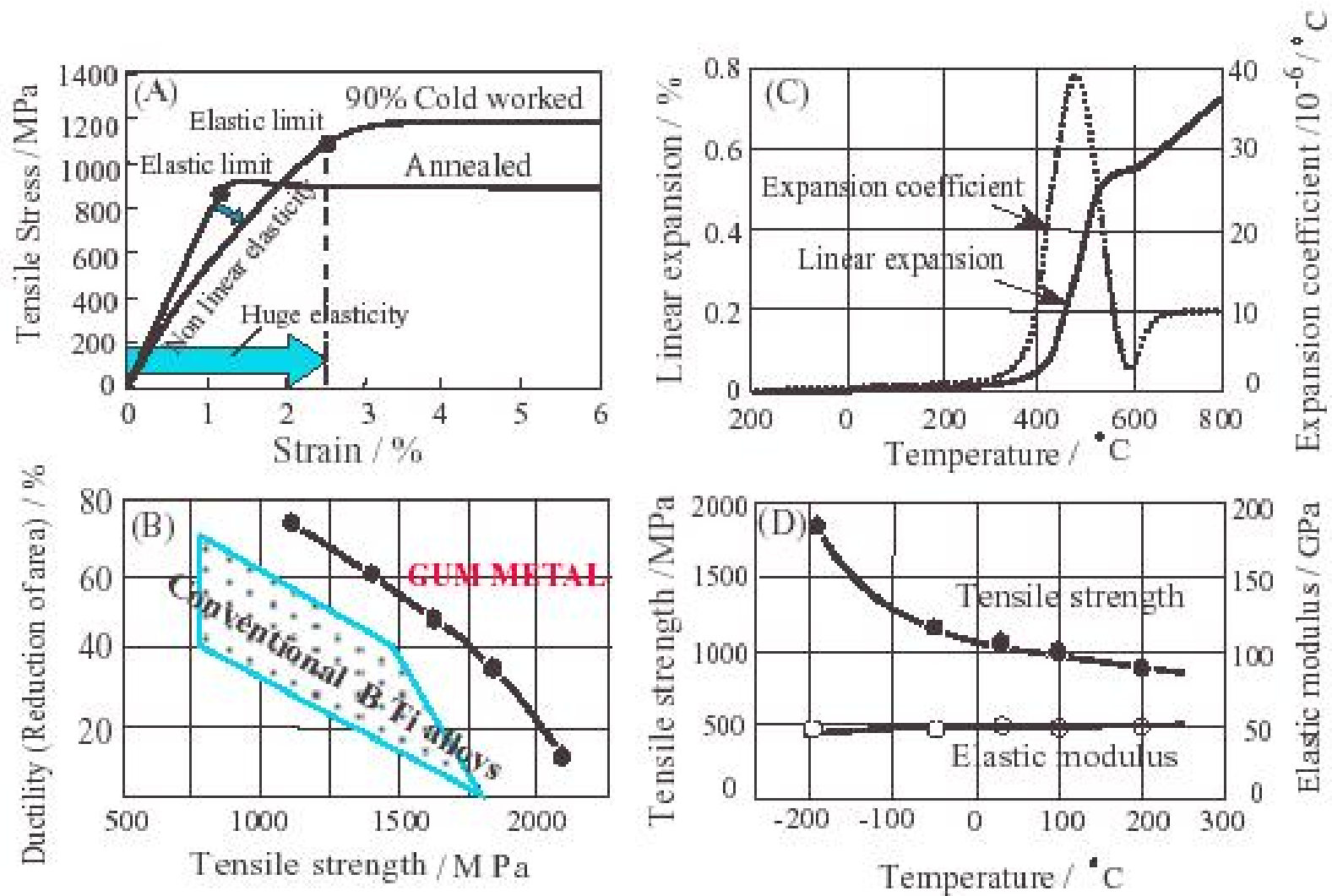


Fig. 3 Anomaly in properties of Ti-Nb-Ta-Zr-O alloys.

Gum Metal (Toyota Ti alloy)



AlBeMet®

AM162

- By weight, contains 62% commercially pure beryllium and 38% commercially pure aluminum
- By Metallurgical definition, AlBeMet® is an alloy but can be considered a composite
- AlBeMet® sheet, plate and bar are powder metallurgy products
 - The powder is produced by a gas atomization process which yields spherical powder with a fine beryllium structure
 - The powder is densified by three consolidation processes, each resulting in different mechanical properties, while maintaining AlBeMet's unique physical properties.

AlBeMet® Property Comparison

Property	Beryllium S200F/AMS7906	AlBeMet AM16H/AMS7911	E-Material E-60	Magnesium AZ80A T6	Aluminum 6061 T6	Stainless Steel 304	Copper H04	Titanium Grade 4
Density lbs/cuin (g/cc)	0.067 (1.86)	0.076 (2.10)	0.091 (2.61)	0.066 (1.80)	0.098 (2.70)	0.29 (8.0)	0.32 (8.9)	0.163 (4.6)
Modulus MSI (Gpa)	44 (303)	28 (193)	48 (331)	6.6 (46)	10 (69)	30 (206)	16.7 (116)	16.2 (106)
UTS KSI (Gpa)	47 (324)	38 (262)	39.3 (273)	49 (340)	46 (310)	76 (516)	46 (310)	96.7 (660)
YS KSI (Gpa)	36 (241)	28 (193)	N/A	36 (260)	40 (276)	30 (206)	40 (276)	86.6 (600)
Elongation %	2	2	< .06	6	12	40	20	20
Fatigue Strength KSI (Gpa)	37.9 (261)	14 (97)	N/A	14.6 (100)	14 (96)	N/A	N/A	N/A
Thermal Conductivity btu/hr/ft/F (W/m-K)	126 (216)	121 (210)	121 (210)	44 (76)	104 (180)	9.4 (16)	226 (391)	9.76 (16.9)
Heat Capacity btu/lb-F (J/g-C)	.46 (1.96)	.373 (1.66)	.310 (1.26)	.261 (1.06)	.214 (.896)	.12 (.6)	.092 (.386)	.129 (.54)
CTE ppm/F (ppm/C)	6.3 (11.3)	7.7 (13.9)	3.4 (6.1)	14.4 (26)	13 (24)	9.6 (17.3)	9.4 (17)	4.8 (8.6)
Electrical Resistivity ohm-cm	4.2 E-06	3.6 E-06	N/A	14.6 E-06	4 E-06	72 E-06	1.71 E-06	60 E-06

VASCOMAX® C-200/C-250/C-300/C-350

 Nominal Mechanical Properties of Small Diameter Bars Following Aging Heat Treatment
 Figure 1

	VascoMax C-200	VascoMax C-250	VascoMax C-300	VascoMax C-350
Ultimate Tensile Strength, psi	210,000	260,000	294,000	350,000
0.2% Yield, psi	206,000	255,000	290,000	340,000
Elongation, %	12	11	11	7
Reduction of Area, %	62	58	57	35
Notch Tensile ($K_t = 9.0$), psi	325,000	380,000	420,000	330,000
Charpy V-Notch, ft-lb	36	20	17	10
Fatigue Endurance Limit (10^6 Cycles), psi	110,000	110,000	125,000	110,000
Rockwell "C" Hardness	43/48	48/52	50/55	55/60
Compressive Yield Strength, psi	213,000	280,000	317,000	388,000

VASCOMAX® C-200
Physical Properties

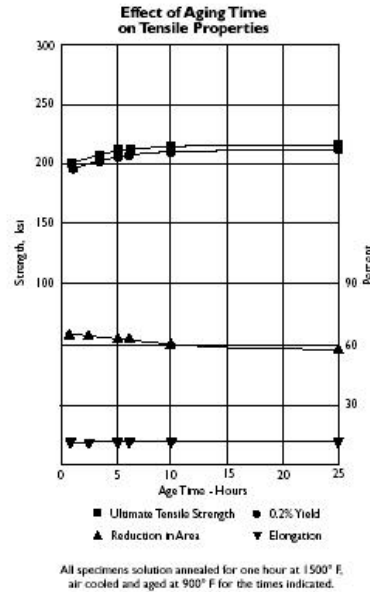
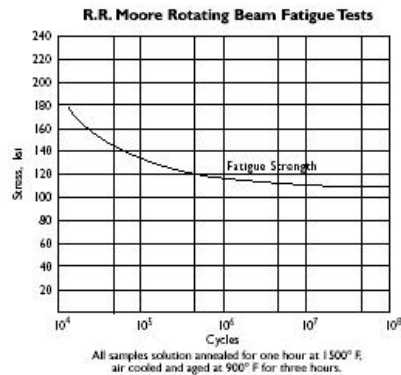
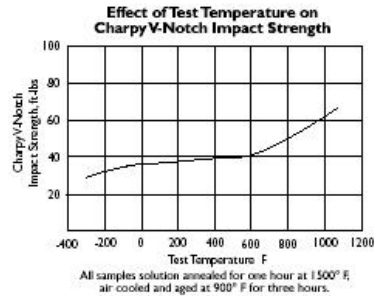
Average Coefficient of Thermal Expansion (70-900° F)	5.6×10^{-6} in/in/°F
Modulus of Elasticity	26.2×10^5 psi
Density	.289 lbs/cu. in. (8.0 g/cc)
Thermal Conductivity at 68° F	11.3 BTU/(ft)(hr)(°F)
at 122° F	11.6 BTU/(ft)(hr)(°F)
at 212° F	12.1 BTU/(ft)(hr)(°F)

Nominal Annealed Properties

Hardness	30 Rc
Yield Strength	100 ksi
Ultimate Strength	140 ksi
Elongation	18%
Reduction of Area	72%

Nominal Room Temperature Properties after Aging

Size	Direction	Hardness Rockwell "C"	Tensile Strength ksi	0.2% Yield Strength ksi	Elongation in 4.5 in. %	Reduction of Area %
5/8" Round	Longitudinal	43.4	212.0	207.7	12.5	61.7
1 1/4" Round	Longitudinal	43.0	214.3	208.5	12.0	60.6
3" Round	Longitudinal	42.8	210.0	204.2	11.9	60.4
6" Square	Longitudinal	43.5	208.4	202.6	11.6	58.8
	Transverse	43.9	206.9	200.1	8.9	41.7
.250" Sheet	Transverse	42.9	218.1	213.0	11.0	45.0

VASCOMAX® C-200


Mechanical Properties**Titanium Ti-6Al-4V (Grade 5), Annealed**

Hardness, Brinell	334	334	Estimated from Rockwell C.
Hardness, Knoop	363	363	Estimated from Rockwell C.
Hardness, Rockwell C	36	36	
Hardness, Vickers	349	349	Estimated from Rockwell C.
Tensile Strength, Ultimate	<u>950 MPa</u>	138000 psi	
Tensile Strength, Yield	<u>880 MPa</u>	128000 psi	
Elongation at Break	14 %	14 %	
Reduction of Area	36 %	36 %	
Modulus of Elasticity	<u>113.8 GPa</u>	16500 ksi	
Compressive Yield Strength	<u>970 MPa</u>	141000 psi	
Notched Tensile Strength	<u>1450 MPa</u>	210000 psi	K_t (stress concentration factor) = 6.7
Ultimate Bearing Strength	<u>1860 MPa</u>	270000 psi	$e/D = 2$
Bearing Yield Strength	<u>1480 MPa</u>	215000 psi	$e/D = 2$
Poisson's Ratio	0.342	0.342	
Charpy Impact	<u>17 J</u>	12.5 ft-lb	V-notch
Fatigue Strength	<u>240 MPa</u>	34800 psi	at 1E+7 cycles. K_t (stress concentration factor) = 3.3
Fatigue Strength	<u>510 MPa</u>	74000 psi	Unnotched 10,000,000 Cycles
Fracture Toughness	<u>75 MPa-m^{1/2}</u>	68.3 ksi-in ^{1/2}	
Shear Modulus	<u>44 GPa</u>	6380 ksi	
Shear Strength	<u>550 MPa</u>	79800 psi	Ultimate shear strength

Electrical Properties**Titanium Ti-6Al-4V (Grade 5), Annealed**

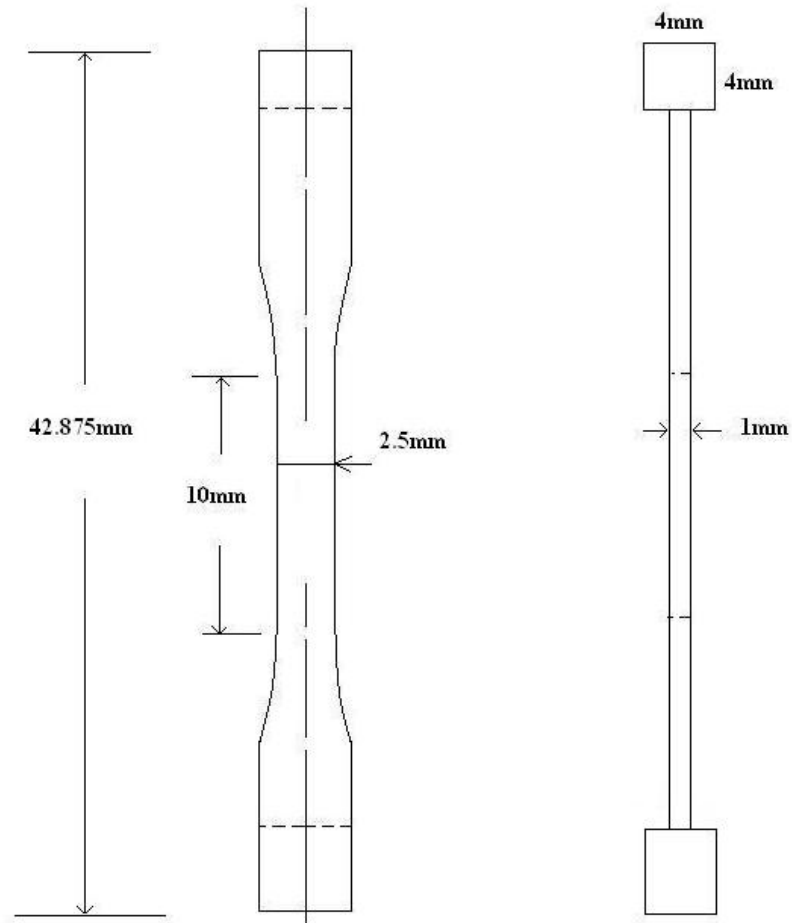
Electrical Resistivity	<u>0.000178 ohm-cm</u>	0.000178 ohm-cm	
Magnetic Permeability	1.00005	1.00005	at 1.6kA/m
Magnetic Susceptibility	3.3e-006	3.3e-006	cgs/g

Thermal Properties

CTE, linear 20°C	<u>8.6 µm/m-°C</u>	4.78 µin/in-°F	20-100°C
CTE, linear 250°C	<u>9.2 µm/m-°C</u>	5.11 µin/in-°F	Average over the range 20-315°C
CTE, linear 500°C	<u>9.7 µm/m-°C</u>	5.39 µin/in-°F	Average over the range 20-650°C
Heat Capacity	<u>0.5263 J/g-°C</u>	0.126 BTU/lb-°F	
Thermal Conductivity	<u>6.7 W/m-K</u>	46.5 BTU-in/hr-ft²-°F	
Melting Point	1604 - 1660 °C	2920 - 3020 °F	
Solidus	<u>1604 °C</u>	2920 °F	
Liquidus	<u>1660 °C</u>	3020 °F	
Beta Transus	<u>980 °C</u>	1800 °F	

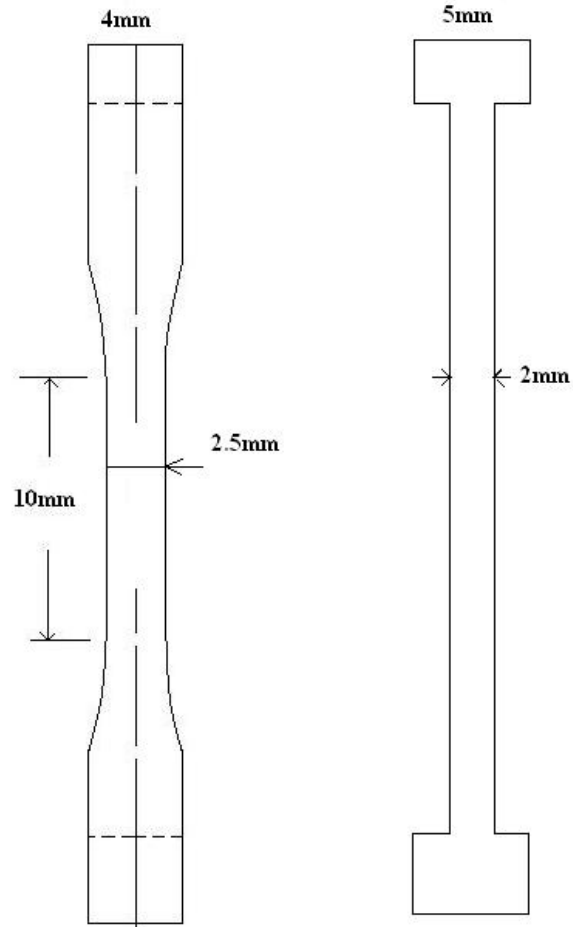
Tensile Specimen

TENSILE SPECIMEN for Vascomax, Gun
Metal and Ti-alloy



Tensile Specimen -- CC

TENSILE SPECIMEN FOR
CARBON-CARBON & AlBeMet



Fracture Specimen

FRACTURE TOUGHNESS
SPECIMEN FOR Vascomax, Ti-alloy
and Gun Metal

